

HSNS

Horizon Sensor for Nano Satellites

ITAR FREE

Space Qualified







Mechanical Interface

	Z CO	— _{Хед} —Х	Center of gravity: Xcg = 1.01 nm Ycg = 0.07 mm Zcg = 21.65 mm Flatness of baseplane: 25 microns Rugosity of baseplane: 0.8 microns
91.78	24.2 48.4	11 X	56.39

Description

Horizon Sensor for **Nano and Micro Satellites** (HSNS) of Solar MEMS is a Quad Thermopile sensor for Earth detection and Nadir vector determination. This device measures the infrared radiation from Space and from Earth **with 4 IR-eyes**, providing accurate and reliable detection and attitude determination.

HSNS is based on previous experience of Solar MEMS making attitude sensors and long research projects on IR sensing devices.

Every HSNS is characterized and **tested and includes a microcontroller for fast** integration with different options like UART or I2C protocols.

Qualification Data and Flight Heritage

Operating	-30° to 70° Celsius
Temperature	30 to 70 Ceisia.

Technical Characteristics

Qualification 30 kRad Total lonizing Dose

Space-grade components
Space qualified microcontroller

Sun sensor	Horizon sensor	
FOV of each IR eye	±2,5°	
Field of View	±64°	
Accuracy	<1 degree, 3sigma (EOL)	
Output rate	10 Hz	
Power supply	5V, 150 mW peak consumption	
Digital interface and connector	UART or I2C, microD 15 pins	
Mechanical interface	90 x 92 x 50 mm	
Mass	120 g	
Housing	Aluminum 6082 (Alodine 1200S and Anodized)	
Orbit	LEO (Customization for different altitudes)	
Lifetime	Designed for 3 years	